[Federal Register: March 11, 2002 (Volume 67, Number 47)]

[CORRECTIONS]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-NM-253-AD; Amendment 39- 12633; AD 2002-02-05]

RIN 2120-AA64

Airworthiness Directives: Airbus Model A300 B2 and A300 B4; A300 B4-600, B4-600R, and F4-600R (Collectively Called A300-600); and Model A310 Series Airplanes

Correction

In rule document 02-2926 beginning on page 6376, in the issue of Tuesday, February 12, 2002, make the following correction:

Sec. 39.13 [Corrected]

On page 6378, in Sec. 39.13, in table 1, under the column" In accordance with", in the fifth line, "A300-29-6004" should read, "A300-29-6003".

[FR Doc. C2-2926 Filed 3-8-02; 8:45 am]

BILLING CODE 1505-01-D

AIRWORTHINESS DIRECTIVE



Aircraft Certification Service Washington, DC

U.S. Department of Transportation Federal Aviation Administration

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The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Title 14 of the Code of Federal Regulations (14 CFR) part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference 14 CFR part 39, subpart 39.3).

CORRECTION

2002-02-05 Airbus Industrie: Amendment 39-12633. Docket 2001-NM-253-AD.

Applicability: Model A300 B2 and A300 B4; A300 B4-600, B4-600R, and F4-600R (collectively called A300-600); and Model A310 series airplanes; certificated in any category; equipped with Dowty or Hamilton Sundstrand ram air turbines (RATs).

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the ram air turbine (RAT) to deploy or operate properly in the event of an emergency, which could result in reduced hydraulic pressure or electrical power on the airplane, accomplish the following:

Overhaul

- (a) Prior to the accumulation of 20 years since the date of manufacture of the airplane, or within 2 years after the date of this AD, whichever occurs later: Overhaul the RAT in accordance with Airbus Service Bulletin A300-29-0118, dated April 20, 2001 (for Model A300 B2 and A300 B4 series airplanes); A300-29-6049, Revision 02, dated September 10, 2001 (for Model A300-600 series airplanes); or A310-29-2087, dated April 20, 2001 (for Model A310 series airplanes); as applicable. Thereafter, repeat the overhaul at least every 20 years, in accordance with the applicable service bulletin.
- **Note 2:** Accomplishment prior to the effective date of this AD of the overhaul in accordance with Airbus Service Bulletin A300-29-6049, dated April 20, 2001, or Revision 01, dated July 23, 2001, is acceptable for compliance with the initial overhaul requirement of paragraph (a) of this AD.
- **Note 3:** The service bulletins identified in paragraph (a) of this AD refer to Hamilton Sundstrand Service Bulletins 730816-29-12, ERPS26T-29-4, and 732365-29-4 as additional sources of service information for the overhaul actions.

Concurrent Modification Requirements

(b) Prior to or concurrently with the overhaul required by paragraph (a) of this AD, perform the applicable modifications specified in the following table:

Table 1Concurrent Modifications					
For model	Modify the airplane by	In accordance with	Which refers to the following additional source of service information:		
(1) A300 series airplanes	(i) Installing a grease nipple and a scraper seal assembly and replacing the locking rod spring with a stronger spring.	Airbus Service Bulletin A300-29-0106, Revision04, dated March 222001.	Hamilton Sundstrand Service Bulletin ERPS26T-29-1.		
	(ii) Replacing the RAT with a modified RAT.	Airbus Service Bulletin A300-29-0115, Revision01, dated June 28, 2000.	Hamilton Sundstrand Service Bulletin ERPS26T-29-2.		
(2) A300-600 series airplanes	(i) Replacing the RAT blade release cable and sheath and modifying the RAT identification plate.	Airbus Service Bulletin A300-29- 6003, dated January 31, 1985, including Change Notice O.A., dated June 9, 1987.	Hamilton Sundstrand Service Bulletin 732365-29-1.		
	(ii) Modifying the RAT	Airbus Service Bulletin A300-29-6005, Revision 1, dated September 2, 1986.	Hamilton Sundstrand Service Bulletin 732365-29-2.		
	(iii) Installing a grease nipple andn a scraper seal assembly and replacing the locking rod spring with a stronger spring.	Airbus Service Bulletin A300-29-6039, Revision04, dated March 22, 2001.	Hamilton Sundstrand Service Bulletin ERPS26T-29-1.		
	(iv) Replacing the RAT with a modified RAT.	Airbus Service A300-29-6046, Revision 02, dated June 28, 2000.	Hamilton Sundstrand Service Bulletin ERPS26T-29-2.		

(3) A310 series airplanes	(i) Reidentifying RATs and RAT assemblies that in good condition, performing functional tests, and reidentifying certain RATs.	Airbus Service A310-29-2003, dated January 20, 1984.	[reserved]
	(ii) Replacing the blade release cablel and sheath and modifying the RAT identification plate.	Airbus Service A310-29-2008, dated January 31, 1985, including Change Notice O.A., dated October 6, 1987.	Hamilton Sundstrand Service Bulletin 730816-29-9.
	(iii) Modifying the RAT	Airbus Service Bulletin A310-29-2011, Revision 1, dated September 2, 1986.	Hamilton Sundstrand Service Bulletin 730816-29-10.
	(iv) Installing a grease nipple and a scraper seal assembly and replacing the locking rod spring with a stronger spring.	Airbus Service Bulletin A310-29-2078, Revision04, dated March 22, 2001.	Hamilton Sundstrand Service Bulletin ERPS26T-29-1.
	(v) Modifying the RAT	Airbus Service Bulletin A310-29-2084, Revision02, dated June 28, 2000.	Hamilton Sundstrand Service Bulletin ERPS26T-29-2.

Note 4: The following Airbus service bulletins are also acceptable for compliance with the applicable requirements of paragraph (b) of this AD:

A300-29-0106, Revision 01, dated September 8, 1997; Revision 02, dated January 26, 1999; and Revision 03, dated June 28, 2000.

A300-29-0115, dated September 14, 1998.

A300-29-6003, dated January 31, 1985.

A300-29-6005, dated June 21, 1985.

A300-29-6039, Revision 01, dated September 8, 1997; Revision 02, dated January 26, 1999; and Revision 03, dated June 28, 2000.

A300-29-6046, dated September 14, 1998; and Revision 01, dated December 16, 1998.

A310-29-2011, dated June 21, 1985.

A310-29-2078, Revision 01, dated September 8, 1997; Revision 02, dated January 26, 1999; and Revision 03, dated June 28, 2000.

A310-29-2084, dated September 14, 1998; and Revision 01, dated December 16, 1998.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 5: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with Secs. 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) The actions shall be done in accordance with Airbus Service Bulletin A300-29-0106, Revision 04, dated March 22, 2001; Airbus Service Bulletin A300-29-0115, Revision 01, dated June 28, 2000; Airbus Service Bulletin A300-29-0118, dated April 20, 2001; Airbus Service Bulletin A300-29-6003, dated January 31, 1985, including Change Notice O.A., dated June 9, 1987; Airbus Service Bulletin A300-29-6005, Revision 1, dated September 2, 1986; Airbus Service Bulletin A300-29-6039, Revision 04, dated March 22, 2001; Airbus Service Bulletin A300-29-6046, Revision 02, dated June 28, 2000; Airbus Service Bulletin A300-29-6049, Revision 02, dated September 10, 2001; Airbus Service Bulletin A310-29-2003, dated January 20, 1984; Airbus Service Bulletin A310-29-2008, dated January 31, 1985, including Change Notice O.A., dated October 6, 1987; Airbus Service Bulletin A310-29-2011, Revision 1, dated September 2, 1986; Airbus Service Bulletin A310-29-2078, Revision 04, dated March 22, 2001; Airbus Service Bulletin A310-29-2084, Revision 02, dated June 28, 2000; and Airbus Service Bulletin A310-29-2087, dated April 20, 2001; as applicable. Revision 1 of Airbus Service Bulletin A300-29-6005 contains the following effective pages:

Page No.	Revision level shown on page	Date shown on page
1, 2	1	Sept. 2, 1986.
3	Original	June 21, 1985.

Revision 1 of Airbus Service Bulletin A310-29-2011 contains the following effective pages:

Page No.	Revision level shown on page	Date shown on page
1, 2	1	Sept. 2, 1986.
3, 4	Original	June 21, 1985.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 6: The subject of this AD is addressed in French airworthiness directive 2001-212(B), dated May 30, 2001.

Effective Date

(f) This amendment becomes effective on March 19, 2002.

Issued in Renton, Washington, on January 30, 2002.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02-2926 Filed 2-11-02; 8:45 am]

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